INTERNATIONAL STANDARD

INTERNATIONAL ORGANIZATION FOR STANDARDIZATION MET MET APODIAR OPTAHUSALUN TO CTAHDAPTUSALUN ORGANISATION INTERNATIONALE DE NORMALISATION

Standard Atmosphere

(identical with the ICAO and WMO Standard Atmospheres from - 2 to 32 km)

Atmosphère Type (identique aux atmosphères standard de l'OACI et de l'OMM entre – 2 et 32 km)

Стандартная атмосфера (от — 2 до 32 км идентична стандартным атмосферам ИКАО и ВМО)

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FOREWORD

ISO (the International Organization for Standardization) is a worldwide federation of national standards institutes (ISO flember Bodies). The work of developing International Standards is carried out mough ISO Technical Committees. Every Member Body interested in a subject for which a Technical Committee has been set up has the right to be represented on that committee. International organizations, governmental and non-governmental, in liaison with ISO, also take part in the work. \bigcirc

Draft International Standards adopted by the Technical Committees are circulated to the Member Bodies for approval before their acceptance as International Standards by the ISO Council. G

International Standard ISO 2533 was drawn up by Technic Committee ISO/TC 20, Aircraft and space vehicles, and circulated to the Member Bodies in April 1972. [The tables of the ISO Interim Standard Atmosphere (see page iii) were circulated separately to the Member Bodies in August 1972 as Addend m 1 and have now been incorporated in the present document.]

It has been approved by the Member Bodies of the following countrie

Austria* Belgium* Brazil Czechoslovakia* Egypt, Arab Rep. of* France* Germany*

India* Ireland* Japan Netherlands* New Zealand* Portugal Romania'

South Africa Thailand' Turkey* United Kingdom U.S.A.* U.S.S.R.*

Also approved Addendum 1.

No Member Body has expressed disapproval of the document.

rerated by FLY-NOTE - The following International Organizations took part in the discussion of this International Standard at all stages of its development :

International Civil Aviation Organization (ICAO).

World Meteorological Organization (WMO).

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The characteristics of the ISO Starter. Successful and interaction of geometric and geopotential altitudes. Sources altitudes were recognized as the most representative when con-current national and international standards and recommendations on use atmosphere [14], [6-7] with the results of recent research. Data from this recent research have been used for calculation of the atmosphere tharacteristics for altitudes from 50 000 to 80 000 m, representing the ISO Interim attracteristic and international standards and recommendations on use tharacteristics for altitudes from 50 000 to 80 000 m, representing the ISO Interim attracteristic and international standards and recommendations on use attracteristic and international standards and recommendations on use attracteristics for altitudes from 50 000 to 80 000 m, representing the ISO Interim attracteristics and attracteristics and attracteristics attra

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Standard Atmosphere

(identical with the ICAQ and WMO Standard Atmospheres from -2 to 32 km)

1 SCOPE AND FIELD F APPLICATION

This International Standard precifies the characteristics of an ISO Standard Atmosphere and is intended for use in calculations and design of flying vehicles, to present the test results of flying vehicles and their components under identical conditions, and to allow unification in the field of development and calibration of instruments. Its use is also recommended in the processing of data from geophysical and meteorological observations.

2 BASIC PRINCIPLES AND CALCULATION FOR-MULAE

2.1 Primary constants and characteristics

The tables of the ISO Standard Atmosphere have been calculated assuming the air to be a perfect gas free from moisture and dust and based on conventional initial values of temperature, pressure and density of the air for mean sea level. The following constants and characteristics are used for calculations and their numerical values are given in table 1 :

 g_n - standard acceleration of free fall. It conforms with latitude $\varphi = 45^{\circ} 32' 33''$ using Lambert's equation of the acceleration of free fall as a function of latitude φ [5]:

$$\begin{array}{l} g_{\varphi} = 9,806 \,\, 16 \,\, (1-0,002 \,\, 637 \,\, 3 \,\cos 2\varphi \\ + \,0,000 \,\, 005 \,\, 9 \,\cos^2 \, 2\varphi) \end{array}$$

- M air molar mass at sea level, as obtained from the perfect gas law (2) when introducing the adopted values ρ_n , ρ_n , T_n , R^* (see table 1);
- N_A Avogadro constant, based on the value of the nuclide ¹²C, atomic mass = 12,000, as adopted in 1961 by the Conference of the International Union of Pure and Applied Chemistry as the basic atomic mass unity;
- p_n standard air pressure;
- R^{*} universal gas constant;
- R specific gas constant;
- ${\it S} \mbox{ and } \beta_{\rm s}$ Sutherland's empirical coefficients in the equation for dynamic viscosity;

- T_o thermodynamic ice-point temperature, at mean sea level;
 - standard thermodynamic air temperature at mean sea level;
 - Celsius ice-point temperature at mean sea level;
 - standard Celsius air temperature at mean sea level;
- $\kappa = \frac{c_p}{c_v}$ adiabatic index, the ratio of the specific heat of air at constant pressure to its specific heat at constant volume;
 - standard air density;

T_

t_o

t_n

 ρ_n

σ

 effective collision diameter of an air molecule; taken as constant with altitude.

| TABLE | 1 – Main constants and characteristics adopted for the |
|-------|--|
| 2 | calculation of the ISO Standard Atmosphere |

| Systipol | Value | Unit of measurement |
|----------------|----------------------------|---|
| g _n | 9,806 65 | m·s⁻² |
| м | 964 420 | kg∙kmol ⁻¹ |
| NA | 602 257 × 10 ²⁴ | kmol ⁻¹ |
| ρ _n | 101,325 × 10 ³ | Ра |
| | 1,013 250 ×10 ³ | mbar |
| | 760 | mmHg |
| R* | 8 314,32 | J · K ⁻¹ · kmol ⁻¹ |
| | | or ►kg · m ² · s ⁻² · K ⁻¹ · kmol ⁻¹ |
| R | 287,052 87 | О . К ⁻¹ · kg ⁻¹ |
| | | or m ² · K ⁻¹ · s ⁻² |
| S | 110,4 | к |
| Τ _o | 273,15 | к |
| r _n | 288,15 | к |
| t _o | 0,00 | °C |
| t _n | 15,00 | °C |
| β _s | 1,458 × 10 ^{⊣6} | $kg \cdot m^{-1} \cdot s^{-1} \cdot K^{-1/2}$ |
| к | 1,4 | dimensionless |
| ρ _n | 1,225 | kg·m ^{−3} |
| σ | 0,365 × 10 ⁻⁹ | m |